



7TH – 10TH MAY 2012  
 BORDEAUX  
 CONVENTION CENTRE  
[www.propulsion2012.com](http://www.propulsion2012.com)



## SP2012 PRELIMINARY PROGRAMME

### Sunday 6th May 2012

|             |                                  |
|-------------|----------------------------------|
| 18:30-19:30 | Registration                     |
| 19:30-20:30 | Welcome Cocktail - PULLMAN HOTEL |

### Monday 7th May 2012

|               |  |
|---------------|--|
| 8:30-9:30     | Registration   |
| 9:30-10:30    | <p><b>ESA-3AF CONFERENCE INTRODUCTION</b></p> <p>M. SCHELLER, President, Association Aéronautique et Astronautique de France (3AF)<br/>         J-J. DORDAIN, Director General, European Space Agency (ESA)<br/>         C. FEICHTINGER, Executive Director, International Astronautical Federation (IAF)<br/>         G. SACCOCCIA &amp; P-G. AMAND, Conference Chairs</p>  |
| 10:30-11:00   | Coffee Break   |
| 11:00 - 12:30 | <p><b>SESSION 1 - ROUND TABLE ON SPACE MISSIONS: MID AND LONG TERM POLICIES</b><br/>         Moderator: P. DE SELDING (SPACE NEWS)</p> <p>Invited Speakers: J-J. DORDAIN, Director General, European Space Agency (ESA )<br/>         M. EYMARD, Director of Launchers, CNES<br/>         S.SCIMEMI , Deputy, International Space Station, Human Exploration and Operations Mission Directorate, NASA<br/>         C.QIUFA, Administrator, CNSA<br/>         S. SCHLECHTRIEM, Director, DLR-Institute of Space Propulsion<br/>         D. PARKER, Director of Technology and Programmes, UK Space Agency<br/>         E.SAGGESE, President, Italian Space Agency</p> |
| 12:30 - 13:30 | Lunch  |





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|               |   |
|---------------|---|
| 13:30 - 15:00 | <p><b>SESSION 2 - AGENCY VIEWS ON PROPULSION FOR LAUNCHERS, SPACECRAFT AND EXPLORATION VEHICLES</b><br/>         Chairs: C. STAVRINIDIS, F. ROSSI (ESA)</p> <p>Invited Speakers: A. FABRIZI, Director of Launchers, European Space Agency (ESA)<br/>         F. ONGARO, Director of Technical and Quality Management (ESA)<br/>         M. EYMARD, Director of Launchers, Centre National d'Etudes Spatiales (CNES)<br/>         B.K. SMITH, Director, Space Flight Systems, NASA Glenn Research Center<br/>         D. THOMAS, Associate Director, Technical, Marshall Space Flight Center<br/>         S. MOSOLOV, Director, Liquid Propulsion Center, Keldysh Research center, Russia</p>  |
| 15:00 - 15:30 | Coffee Break  |
| 15:30 - 17:30 | <p><b>SESSION 3 - NEXT STEPS IN SPACE PROPULSION: AN INDUSTRIAL AND OPERATOR PERSPECTIVE</b><br/>         Moderator: J.P. TORTORA (EUROSPACE)</p> <p>Invited Speakers : L. LAURENT, Director of programmes, ARIANESPACE<br/>         D BAIR, Chief Technical Officer, EUTELSAT Communications<br/>         H.GILIBERT, Chief Technical Officer, Astrium Space Transportation<br/>         C. LAZARO, Deputy Technical Officer, Thales Alenia Space<br/>         H.AUSTRUY, Chief Executive Officer, Safran-SME<br/>         A. ROHNE, Director of Propulsion, Astrium GmbH – Space Transportation<br/>         R. MEYERS, General Manager, Aerojet Redmond-Operations<br/>         K. ISHII, President of IHI Aerospace<br/>         J.HENNIG, President, Moog Inc, Space &amp; Defense Group<br/>         V. MURASHKO, Director and General Designer of the Experimental Design Bureau "Fakel", Kaliningrad<br/>         J.YATES, Strategy Board, UK Government<br/>         P.G.LASAGNI, Chief Executive Officer, Avio Space<br/>         M. SION, Vice President Space Engines, SNECMA</p> |
| 19:30 - 23:30 | <p>Cocktail - Château Sogéant<br/>         30 route de Cénac BP 7 - 33880 Saint Caprais de Bordeaux - Tel : 05.56.20.71.20</p>  |



### SP2012 PRELIMINARY PROGRAMME

#### Tuesday 8th May 2012

| KEYNOTE SPEECHES<br>Technologies for Space Exploration<br>G.SACCOCCIA, Head of Propulsion & Aerothermodynamics, ESA-ESTEC |  |   |   |  |  |   |
|---|--|---|---|--|--|---|
| Coffee Break  |  |   |   |  |  |   |
| Room  | SC1  | SC2   | SC3   | ST1  | ST2  | Amphi C   |
|   | Session 1 - SC - Electric Propulsion Systems (Hall Effect Thrusters)   | Session 2 - SC - Propulsion Components - Chemical   | Session 3 - SC - Green Propulsion for Spacecraft  | Session 4 - Space Transportation Roadmaps  | Session 5 - ST - Solid propulsion 1  | Session 6 - Overview of Current Programmes - Agencies   |
| Chairmen  | M.Andreucci (ALTA)<br>M.Patterson (NASA)   | R.Voeten (MOOG-Bradford)<br>N.Goody (ESA)   | K.Anflo (ECAPS)<br>M.Smith (ESA)  | P. Fortunier (CNES)<br>K. Okita (Jaxa)   | D. Boursy (SPS)<br>G. Vigier (Astrium)   | P-G. Amand (SAFRAN-SME/3AF)<br>M.F. Rossi (ESA)   |
| 10:30 - 10:50   | Alta's HT-400<br>D. DIGNANI - Alta SPA   | Sentinel-3 propulsion subsystem calibrated trimming orifice design and qualification<br>F. LAVERTY - Thales Alenia Space              | Replacement of conventional spacecraft propellants with green propellants<br>F. VALENCIA BEL - ESA  | Enabling a national engine to meet the needs of future multi-stage in-space applications<br>W. LEAHY - NASA      | Variations of solid rocket motor preliminary design for small TSTO launcher<br>E. DUMONT - DLR   | NASA Technology roadmap for in-space propulsion technology<br>L. MICHAEL MEYER - NASA                       |
| 10:50 - 11:10   | Performances of a variable channel width hall thruster using xenon and krypton<br>S. MAZOUFFRE - ICARE, CNRS   | Experimental investigation of the catalyst life characteristics of the 4N hydrazine monopropellant thruster<br>G. FUJII - JAXA        | Regeneratively cooled liquid oxygen/methane technology development between<br>C. GREENE - UTRC  | H-IIA upgrade – Evolving plan of Japanese primary launch system<br>M. ATSUMI - Space Systems                     | Overview of large solid rocket motor solutions for new generation launcher<br>A. COPEY - Safran SPS  | Overview of chemical propulsion activities<br>M. FORD - ESA-ESTEC   |
| 11:10 - 11:30   | PPS NG Hall effect thruster addressing extended propulsive mission<br>V. VIAL - Snecma-Safran  | Feasible study on ignition system candidates for HAN based propellant<br>T. IZUKA - Tokyo Metropolitan University                     | Determination of fluid properties of the green propellant FLP-106 and related material and component testing with regard to applications in space missions<br>H. CIEZKI - DLR | Concept Study of Japan's Next Flagship Launch System, H-X<br>K. OKITA - JAXA                                     | Five-segment solid rocket motor development status<br>A. PRISKOS - NASA-MSFC   | Galileo FOC OCS: A joint US-European approach to monoprop systems for constellations<br>M. MARCHIONNI - ESA |
| 11:30 - 11:50   | Next generation of thruster module assembly (TMA-NG) : advanced solution for electric orbit raising and station keeping<br>A. LORAND - Snecma-Safran | Experimental study of cryogenic swirl flow at sub to supercritical condition<br>S. CHO - Seoul National University                    | Two years of in-space demonstration and qualification of an ADN-based propulsion system on prisma<br>K. ANFLO - ECAPS   | NASA's Launch propulsion systems technology roadmap<br>P. MC CONNAUGHEY - NASA                                   | VEGA Solid stages post-static firing tests reconstruction : sensitivity analysis and data correlation<br>E. CAVALLINI - University of Rome | Overview of NASA's in-space propulsion technology program for future science missions<br>D. ANDERSON - NASA |
| 11:50 - 12:10   | Measurements of plasma properties in the plume of a hall effect thruster<br>K. DANNENMAYER - ICARE, CNRS   | MON Propellant tank loading, decontamination and reloading<br>S. MURPHY - Thales Alenia Space   | Development of a novel green propellant propulsion system<br>M. DRUBE - Surrey Satellite Technology   | Technology maturation in the frame of the FLPP high thrust liquid propulsion programme<br>T. KACHLER - ESA-ESTEC | Very large monolithic motor project for European future launcher family<br>D. BOURY - SPS  | ESA Electric propulsion activities<br>J. GONZALEZ DEL AMO - ESA   |
| 12:10 - 12:30   | VUV radiation in a hall effect thruster used for the positioning of geostationary satellites<br>L. DA SILVA - Instituto Superior Technico            | Development and qualification of a silica free diaphragm for space propellant tanks<br>R. BELLAROSA - MT Aerospace Satellite Products | Suggestion for the unified specific impulse as an alternative selection criterion for green propellants<br>W. BOUJILJA - DLR  | Main tendency of strategy of NPO energomash in current conditions<br>V.S. SUDAKOV - NPO Energomash               | Overview of safran activities on vega launcher<br>D. BOURY - SPS   | Current and planned activities for electric propulsion<br>N. PUETTMANN - DLR                                |
| 12:30 - 14:00   | Lunch  |   |   |  |  |   |

SP2012 PRELIMINARY PROGRAMME

Tuesday 8th May 2012

| Room          | SC1   | SC2   | SC3  | ST1  | ST2   | Amphi C  |
|---------------|---|---|--|--|---|--|
|               | Session 7 - SC - Electric Propulsion Systems (Helicon and Hollow Cathode Thrusters)   | Session 8 - SC - Propulsion Components - Electric   | Session 9 - SC - Modelling - Chemical Propulsion   | Session 10 - ST - Liquid propulsion Turbomachineries   | Session 11 - ST - Nozzle Extensions and Thrust Vectoring Control  | Session 12 - Overview of Current Programmes  |
| Chairmen      | D.Pavarin (Padua University)<br>C.Edwards (ESA)   | G.Matticari (Selex Galileo)<br>J.Gonzalez del Amo (ESA)   | J.Muylaert (Von Karman Institute)<br>M. Ford (ESA)   | F. Jean (Snecma)<br>N. Ierardo (ESA)   | U. Palmnas (VOLVO)<br>E. Gautronneau (SPS)  | G.Saccoccia (ESA)<br>N. Manesse (Snecma)   |
| 14:00 - 14:20 | The effect of plasma source surface area on helicon double layer thruster performance<br>T. HARLE - University of Surrey  | Small GEO xenon propellant supply assembly : pressure regulation panel test results and comparison with EcosimPro predictions<br>S. NACLERIO - IberEspacio  | Numerical modeling of the thermal behavior of rafaél hydrazine thrusters<br>L. APPEL - Rafael  | The cavitating pump rotordynamic test facility at Alta SPA : upgraded capabilities of a unique test rig<br>L. TORRE - Alta SPA | Thrust shock vector control of an axisymmetric rocket nozzle via transverse gas injection<br>V. ZMJANOVIC - ICARE, CNRS                               | A study for Mars manned exploration<br>D. DORNEY - NASA-MSFC   |
| 14:20 - 14:40 | Design, development and characterization of the helicon plasma thruster of the EU FP7 HPH.com program<br>D. PAVARIN - University of Padua   | A new xenon flow control unit for micropropulsion<br>P. HARMANN - AST   | Improvement and validation of low g sloshing modelling for Spacecraft applications<br>B. BUSSET - Astrium  | Overview of the propellant electric pump (ISP-1 project)<br>A. LEMAITRE - Snecma   | VINCI Engine composite nozzle extension. Development status after the first test firing campaign<br>T. PICHON - SPS                                   | LISA Pathfinder micropropulsion subsystem : an overview of available European micropropulsion options<br>P. BIANCO - Astrium             |
| 14:40 - 15:00 | RF hollow cathode microthruster : simulation, plasma parameters and thrust measurements<br>C. CHARLES - Australian National University  | Weldless liners for high pressure storage COPV<br>F. FERNANDES - Omnidea, Lda   | Experimental characterization of the priming phase using a propellant line mock-up<br>M. LEMA - Von Karman Institute   | Compressible multi-phase and multi-scale cavitation model<br>D. LIUZZI - University of Rome                                    | The SWAN Vulcain 2 sandwich nozzle program, an option for improving ariane<br>L. BROX - Volvo Aero Corporation  | Development and qualification of the BepiColombo electric propulsion system<br>H. GRAY - Astrium   |
| 15:00 - 15:20 | A reduced-order model of a low-current orificed hollow cathode* and it will present a newly developed numerical model for the performance evaluation of hollow cathodes for Hall thruster applications<br>R. ALBERTONI - Alta SPA | Interaction of pressurised xenon with elastomeric seals<br>G. MORRIS - Astrium  | Combustion mechanism of binary mixtures with energetic binder and active or inert filler<br>S.A. RASHKOVSKIY - Institute for Problems in Mechanics RAS       | High-speed permanent magnet motor to drive cryogenic pump<br>L. DLUGIEWICZ - Mikroma   | More electric launchers. A status of SRM electro-mechanical thrust vector control systems (EMTVC)<br>D. DESCAMPS - SABCA                              | High power electric propulsion technology for space exploration (Hiper) : results and consolidation of scenarios<br>T. MISURI - Alta SPA |
| 15:20 - 15:40 | Completion of HET and RIT characterization with atmospheric propellants<br>D. DIGNANI - Alta SPA  | Fibre reinforced titanium matrix composites for xenon propellant tanks<br>S. KYLE-HENNEY - TISICS   | Model of Ignition of conventional and nano aluminum particles covered with aluminum oxide film<br>S.A. RASHKOVSKIY - Institute for Problems in Mechanics RAS | Set-up and testing activities at the FAST3 test stand for LCH4 turbopumps<br>N. IERARDO - AVIO SPA                             | Thrust vector control flight preparation and flight data exploitation<br>T. VANTHUYNE - SABCA   | Prospects of dvina-TM unified transport module with electric propulsion<br>V.G. PETUKHOV - RIAME   |
| 15:40 - 16:00 | A hollow cathode thruster performance model<br>D. FROLLANI - Mars Space   | Modular Multi-application Electric Propulsion Diagnostic Package (MM-EPDP) for advanced characterization of the Electric Propulsion – spacecraft interactions<br>G. MATTICARI - Thales Alenia Space | Validation program for propulsion system propellant tubing<br>R. KILLINGER - Astrium   | Space propulsion applications for high enthalpy solid hydrogen gas generation<br>P. YVART - SME Safran                         | Simulation of the dynamic response of the electro-mechanical actuators of SRMs during the ignition transient application to P80<br>E. BANDELIER - SPS | The FP7 SPARTAN Program status and achievements<br>G. PARISENTI - Thales Alenia Space  |
| 16:00 - 16:30 | Coffee Break  |   |  |  |   |  |

## SP2012 PRELIMINARY PROGRAMME

### Tuesday 8th May 2012

| Room   | SC1   | SC2   | SC3  | ST1   | ST2   | Amphi C   |
|--|---|---|--|---|---|---|
|  | <b>Session 13 - SC - Electric Propulsion Systems (Ion Thrusters)</b>  | <b>Session 14 - SC - Cold Gas and Chemical Propulsion Systems</b>   | <b>Session 15 - SC - Modelling - Electric Propulsion</b>   | <b>Session 16 - ST - Solid propulsion 2</b>   | <b>Session 17 - ST - Liquid propulsion Combustion Chambers and Cooling Technics</b>   | <b>Session 18 - Overview of Current Programmes - Programmes and Companies</b>                         |
| Chairmen   | R.Killinger (ASTRIUM)<br>B.Fallis (ESA)   | D.Gale (ASTRIUM)<br>M.Ford (ESA)  | L.D'Agostino (ALTA)<br>J.Longo (ESA)   | M. Biagioni (Avio)<br>E. Robert (CNES)  | O. Haidn (DLR)<br>M. Atsumi (MHI)   | E. Abriat (Moog)<br>A. De Lillis (ASI)  |
| 16:30 - 16:50  | Performance and efficiency study of miniaturized inductive coupled ion thrusters<br>H. NEUMANN - IOM Leipzig                          | Production and acceptance of the GAIA cold gas micro propulsion system : status of flight hardware activities and outlook on near future perspectives<br>G. MATTICARI - Thales Alenia Space | Particle In cell simulation of FEEP thruster plume based on the PICPlus code<br>L. PAITA - Alta SPA  | Preparation, characterization and thermal stability of gamma alane AIH3<br>Y. BATONNEAU - LACCO, University of Poitiers       | Design and testing status of closed cycle combustion devices within the future launcher preparatory programme<br>R. STRUNZ - Astrium  | Aerojet propulsion for next generation GEO comsat platforms<br>C. FRED WILSON - Aerojet               |
| 16:50 - 17:10  | Operation of the PEGASES thruster with Xe<br>S. MAZOUFFRE - ICARE, CNRS   | Engineering validation model for the Exomars bi-propellant propulsion subsystem<br>S. PAVON - OHB-System AG   | Electric propulsion plume impingement studies tools for GEO satellites<br>A. DEMAIRE - OHB Sweden  | Validation of a new numerical platform CEDRE for the simulation of solid rocket motor ignition transient<br>T. PEVERGNE - SPS | Influence of coolant flow direction on flowfield and heat transfer characteristics in a regeneratively cooled thrust chamber<br>H. NEGISHI - Japan Aerospace Exploration Agency | OHB System's propulsion achievements. Heritage and prospects<br>M. PEUKERT - OHB System AG            |
| 17:10 - 17:30  | Bepi-Colombo high power propulsion electronics challenges and performances<br>J. PALENCIA - Astrium                                   | Electrolytic decomposition of HAN monopropellant in microscale of Alumina-PDMS composite<br>K.S. KOH - University of Nottingham Malaysia Campus   | Simulation of interactions between spacecraft and electric thrusters using the SPIS tool<br>M. WARTELSKI - Astrium   | Iron oxide catalysts for composite solid rocket propellants<br>A. REINA - Politecnico di Milano                               | Flowfield and heat transfer characteristics of a GH2/LOX calorimeter chamber with multi-injector elements<br>Y. DAIMON - Japan Aerospace Exploration Agency                     | Propulsion systems for Boeing's CST-100 Spacecraft<br>T. LORIER - Pratt & Whitney Rocketdyne          |
| 17:30 - 17:50  | Investigation of an RF ion source characteristics with discharge chambers made of different materials<br>H. LOEB - Giessen University | Design and development of a dual mode bipropellant propulsion system for the European student moon orbiter<br>L. FERRARIO - Politecnico di Milano   | Simulation of the interactions between a hall thruster and the Dubai-Sat 2 low earth orbit satellite using SPIS<br>J.Y. LEE - Korea Advanced Institute of Science and Technology | Internal ballistic model with erosion characterization for high performance SRMS<br>E. CAVALLINI - University of Rome         | Prediction of wall thermal behavior in regeneratively-cooled thrust chambers<br>M. PIZZARELLI - University of Rome  | Development and progress of a long life 1N monopropellant hydrazine thruster<br>N. SOLWAY - AMPAC ISP |
| 17:50 - 18:10  | Development, integration and test of a power control unit for mini-RIT ion thruster<br>L. CERUTI - SELEX GALILEO SPA                  | ESMO Cold gas propulsion system<br>D. GIBBON - Stuttgart University   | Experimental investigation of two interacting thruster-plumes downstream of the nozzles<br>A. HOLZ - DLR   | Experimental observation of carbon composite ablation under impact of alumina droplets<br>P. FUZET - CNES-SPCTS               | Accelerating fluid-structure interaction analyses applied to cooled rocket thrust chambers<br>D.S.C. KOWOLLIK - Institute of Aircraft Design and Lightweight Structures         | SGEO Development status and electric propulsion subsystem overview<br>M. DE TATA - OHB System AG      |
| <b>14:00 - 18:00 ESPSS Overview &amp; Workshop Chaired by J. Steelant (ESA), Francesco Di Matteo (ESA) and Marco De Rosa (ESA)</b> |   |   |  |   |   |   |

### SP2012 PRELIMINARY PROGRAMME

#### Wednesday 9th May 2012

| KEYNOTE SPEECH   |   |  |  |  |  |  |
|--|---|--|--|--|--|--|
| Taking the Next Step: Innovative Uses of Nuclear Energy for Space<br>S.HOWE, Director, Center for Space Nuclear Research |   |  |  |  |  |  |
| Room   | SC1   | SC2  | SC3  | ST1  | ST2  | Amphi C  |
|  | Session 19 - SC - Electric Propulsion Systems (PPTs and Electrothermal Thrusters)   | Session 20 - SC - Chemical Propulsion Systems - General  | Session 21 - SC - Modelling - Electric Propulsion  | Session 22 - ST - Liquid propulsion Combustion 1   | Session 23 - ST - Solid propulsion 3 & Pressure oscillations   | Session 24 - ST - Hypersonic Propulsion  |
| Chairmen   | R.Meyers (Aerojet)<br>M.Andrenucci (ALTA)   | J.M.Autric (Astrium)<br>C.Hunter (ESA)   | E.Ahedo (Madrid University)<br>M.Martinez Sanchez (MIT)  | M. Habiballah (Onera)<br>G. Langel (Astrium)   | P Cloutet (SAFRAN SME)<br>E. Robert (CNES)   | M. Sippel (DLR)<br>S. Triqueneaux (Air Liquide)  |
| 09:00 - 09:20  | Effect of Inductive Coil Geometry on the Performance of a Pulsed Inductive Plasma Thrusters<br>A.K. HALLOCK - NASA                        | Description of the propulsion architecture for the Exomars EDL Demonstrator Module (EDM)<br>F. DE DINECHIN - Thales Alenia Space | Advanced simulation of magnetic nozzles for plasma thrusters<br>E. AHEDO - Universidad Politécnica de Madrid                 | Thermodynamic analysis of liquid core presence in supercritical injection<br>D. BANUTI - DLR   | Ariane 5 SRM pressure and thrust oscillations<br>A. ESNAULT - SPS  | Design of the gas sub-system for LEA<br>S. TRIQUENEUX - Air Liquide  |
| 09:20 - 09:40  | Ablative pulsed plasma thruster for the small satellite " Soyuz-Sat-O "<br>V. TYUTIN - RIAME  | Further development of the ATV 200N bipropellant thruster<br>S. KRAUS - Astrium  | Hall thruster lifetime modeling with an axisymmetric PIC model<br>W. CHOI - Tech-X Corporation                               | Validation of turbulent combustion models for 3D-Simulations of liquid H2-O2 rocket combustors<br>B. IVANCIC - Astrium                           | Reduction of SRM pressure oscillations through the use of bevelled fuel grains<br>J. ANTHOINE - ONERA  | Design and optimization of a small scale M=8 scramjet propulsion system<br>T. LANGENER - ESA-ESTEC                                 |
| 09:40 - 10:00  | Assessment of the plasma characteristics during the discharge of an ablative pulsed plasma thruster<br>T. SCHÖNHERR - University of Tokyo | Exomars trace gas orbiter reaction control system<br>R. LESCOUZERES - OHB System AG  | Three-dimensional fully kinetic particle-in-cell model of hall-effect thruster<br>F. TACCOGNA - CNR                          | Large eddy simulation of bluff-body stabilised jets and nonpremixed flames<br>N. SALVADOR - INPE   | Segmented SRM pressure oscillation demonstrator-POD-X program<br>E. GAUTRONNEAU - SPS  | Development of high-speed airbreathing propulsion technology for application in space transportation systems<br>F. FALEMPIN - MBDA |
| 10:00 - 10:20  | Development of 125 mN and 250 mN Resisto-jet Thrusters<br>A. PASSARO - Alta SPA   | IXV Reaction control system<br>O. MAILLAN - Thales Alenia Space  | FEM Analysis of hall thrusters<br>F. SHAFIEI - Sharif University of Technology   | Experimental flashing jet thermal characterization by non intrusive optical techniques<br>M.R. VETRANO - Von Karman Institute for Fluid Dynamics | SRM Aero-acoustics resonance numerical simulation by means of Q1D model<br>E. CAVALLINI - University of Rome   | Pre-cooler design and development for combined-cycle engines<br>R. BOND - Reaction Engines   |
| 10:20 - 10:40  | Design and performance study of a low thermal inertia resistojet model<br>A. KOBIERA - Warsaw University of Technology                    | Progress towards realisation of a 1.1 kN apogee engine for interplanetary propulsion<br>L. NAICKER - Ampac-ISP                   | Model for predicting the performance of a hall effect thruster fed with non-conventional propellants<br>T. MISURI - Alta SPA | Identification of high temperature dusted flows by means of local disturbance method<br>I. KUZNETSOV - Baltic State Technical University         | An experimental investigation of relationship between local burning rate and isochrone surface during slurry casting process of solid propellant grains<br>H. HASEGAWA - NOF Corporation | Injector layout optimization for the LAPCAT MR2 Mach 8 Cruiser<br>J.J. VELLARAMKALAYIL - University of Stuttgart                   |
| 10:40 - 11:10  | Coffee Break  |  |  |  |  |  |



SP2012 PRELIMINARY PROGRAMME

Wednesday 9th May 2012

| Room          | SC1   | SC2   | SC3  | ST1   | ST2  | Amphi C  |
|---------------|---|---|--|---|--|--|
|               | Session 25 - SC - Propulsion Components - Electric  | Session 26 - SC - Chemical Propulsion Systems - Flight Testing and Experience   | Session 27 - SC - Advanced Propulsion Systems  | Session 28 - ST - Liquid propulsion Stages  | Session 29 - ST - Hybrid propulsion 1  | Session 30 - ST - Launcher's Attitude Control Systems 1  |
| Chairmen      | V.Kim (Riame)<br>S.Mazouffre (ICARE-CNRS)   | M.Pessana (Thales Alenia Space)<br>D.Gibbon (SSTL)  | L.Johnson (NASA)<br>E.Gengembre (ESA)  | S. Bianchi (Air Liquide)<br>K. Schäffer (DLR)   | P. Gautier (SAFRAN-SME)<br>M. De Rosa (ESA)  | N. Girard (CNES)<br>R. Strunz (Astrium)  |
| 11:10 - 11:30 | MEMS Flow control module for Ion engines<br>J. BEJHED - NanoSpace   | Japanese new 1N monopropellant thruster qualification test results<br>H. IKEDA - IHI Aerospace  | Electric solar wind sail : current status, flight testing and roadmap<br>P. JANHUNEN - Finnish Meteorological Institute                        | Cryogenic propulsion stage configuration in support of NASA's multiple design reference missions<br>S. HANNA - NASA                                 | An updated roadmap for hybrid propulsion<br>S. HENRY - SME   | Experimental and analytical demonstration of increased temperature limits for the hydrazine propelled roll and attitude control for the VEGA launcher<br>V. BOMBELLI - Astrium |
| 11:30 - 11:50 | Influence of borosil composition of the discharge chamber on the HET performance<br>J. IBARZO - Tecnalia                          | Qualification tests results for 4N thrust long life monopropellant thrusters<br>T. NAGAO - IHI Aerospace  | Electric sail propulsion for asteroid touring mission<br>U. KVELL - Tartu Observatory  | A simplified model for upper stage performance parameter calculation<br>E. ROSENKO - Astrium  | Design and development at ONERA of a hybrid rocket motor<br>P. PREVOT - ONERA                          | GRASP : Status and future of green propellants<br>C. SCHARLEMANN - University of Applied Sciences Wiener Neustadt  |
| 11:50 - 12:10 | Investigation of the SPT discharge chamber erosion rate dependence on thruster operating mode<br>V. KIM - RIAME                   | Verification test results of Astrium's 22N Non-ITAR bipropellant thruster for lunar lander application<br>S. KRAUS - Astrium  | Propulsion using plasma dynamics<br>L. JOHNSON - NASA  | Development of a new generation of space liquid rocket engines, on the basis of a pneumopump propellant supply system<br>V. SHNYAKIN - Yuzhnoye SDO | The development of the hybrid rocket for bloodhound SSC<br>D. JUBB - The Falcon Project                | Conceptual study of a HPGP reaction control system for a launcher upper stage<br>D. WELBERG - Astrium  |
| 12:10 - 12:30 | A new active control method for power processing unit of a 20mN class hall thruster<br>H. OSUGA - Mitsubishi Electric Corporation | Qualification of alphas platform chemical propulsion system<br>M. THERIC - Astrium  | Augmentation of solar thermal propulsion systems via phase-change thermal energy storage and thermal-electric conversion<br>D.B. SCHARFE - ERC | Self-pressurizing saturated propellant feed systems with heat transfer<br>P. TIAGO - Omnidea  | Numerical Simulations of an H2O2 Vortex Hybrid Rocket Motor<br>F. BARATO - CISAS, University of Padova | HPGP Thruster demonstrator development for launcher upper stage reaction control system<br>K. ANFLO - ECAPS  |
| 12:30 - 12:50 | Power processing unit activities<br>E. BOURGUIGNON - Thales Alenia Space  | Feasibility study and on-ground demonstration test results of electric-motor powered centrifugal pump-fed monopropellant propulsion system<br>A. TOKAJI - IHI Aerospace | Introducing the magnetic field reconnection<br>G. PANNASENTI - Thales Alenia Space   | Staged combustion cycle rocket engine design trade-offs for future advanced passenger transport<br>M. SIPPPEL - DLR                                 | Technology demonstrators for hybrid rocket propulsion in Europe<br>S. SOLLER - Astrium                 | Thermal or catalytic decomposition of energetic ionic liquids. Can we get oxygen?<br>C. KAPPENSTEIN - LACCO, University of Poitiers  |
| 12:50 - 14:00 | Lunch   |   |  |   |  |  |

SP2012 PRELIMINARY PROGRAMME

Wednesday 9th May 2012

| INTERNATIONAL GREEN PROPULSION PANEL |  |  |  |  |  |  |
|--------------------------------------|--|--|--|--|--|--|
| 14:00 - 15:30                        |  |  |  |  |  |  |
| 15:30 - 16:00                        | Coffee Break   |  |  |  |  |  |
| Room                                 | SC1  | SC2  | SC3  | ST1  | ST2  | Amphi C  |
|                                      | Session 31 - SC - Electric Propulsion (Cubesat and Nanosat Applications)   | Session 32 - SC - Propulsion Components - General  | Session 33 - SC - Modelling - Chemical Propulsion (Focus on ECOSIMPRO)   | Session 34 - ST - Liquid propulsion Cooling Technics   | Session 35 - ST - Hybrid propulsion 2  | Session 36 - ST - Liquid propulsion. Mechanical and Life   |
| Chairmen                             | B.Sanders (TNO)<br>J.Muylaert (VKI)  | A.Demaire (OHB Sweden)<br>S.Hyde (ESA)   | M.Onofri (University of Rome)<br>M.De Rosa (ESA)   | M. Meyer (NASA)<br>P. Alliot (SPS)   | L. Galfetti (Polimi)<br>H. Blanchard (SAFRAN-SME)  | P. Danous (Sneema)<br>T. Kachler (ESA)   |
| 16:00 - 16:20                        | An electric propulsion module for cubesats<br>C. SCHARLEMANN - Fotec   | Demonstration of new ultrasonic flowmeter propulsion applications<br>R. MATTHIJSSEN - Bradford Engineering   | EcosimPro numerical simulations for Exomars EDM propulsion design and operational analyses<br>F. LAVERTY - Thales Alenia Space     | Simulation of liquid and transcritical cooling films in rocket combustion chambers<br>C. HOEGLAUER - Astrium                                     | Modeling of metallized solid fuel hybrid propulsion system<br>A. MAKLED - MTC  | Progress and perspectives in the life duration prediction of liquid rocket engines components<br>Presented by A. DU TERPRE - Sneema            |
| 16:20 - 16:40                        | A $\mu$ PPT for Nano-satellite application : Design and preliminary experimental results<br>A. MINGO PEREZ - University of Southampton | He and Xe electronic regulator preliminary sizing for UPS and PPS propulsion systems on satellite. Comparison with mechanical<br>M. LYSZYK - Thales Alenia Space | Adaptation of the ESPSS/ECOSIMPRO platform for the design and analysis of liquid propellant rocket engines<br>J. MORAL - EcosimPro | Experiments on film cooling in a GOX/Kerosene rocket combustion chamber<br>C. KIRCHBERGER - Technische Universität München                       | Labscale characterization of solid fuels for hybrid propulsion<br>L. GALFETTI - Politecnico di Milano                                | High cycle fatigue testing of structural materials in high pressure hydrogen atmosphere<br>M. BRUCHHAUSEN - JRC-IET                            |
| 16:40 - 17:00                        | MicroThrust MEMS electro spray emitters – integrated microfabrication and test results<br>C. RYAN - Queen Mary University of London    | Study of electronic pressure regulator for telecommunications satellite applications<br>J. STANOJEV - OHB Sweden   | Analysis of bepicolombo dual-mode propulsion system through an EcosimPro model<br>V. DA RONCO - Astrium                            | Experimental investigation and CFD-simulation of the film cooling in O2/CH4 subscale combustion chamber<br>D. SUSLOV - DLR                       | Chemically modified and nano-composite polymeric substrates as novel fuels for hybrid propulsion<br>P. JOSEPH - University of Ulster | Influence of anisotropic damage on the lifetime prediction of regeneratively cooled nozzle structure<br>V. TINI - RWTH Aachen University       |
| 17:00 - 17:20                        | Breadboard development & test of a micro system technology electric propulsion system<br>B. SANDERS - TNO                              | Satellite propellant volume gauging using gas injection<br>P. RANGSTEN - NanoSpace   | Effects of a satellite mission on the propulsion subsystem<br>C. KOPPEL - Kopoos Consulting  | Experimental study of methane heat transfer characteristic in regenerative cooling channels<br>H. KAWASHIMA - Japan Aerospace Exploration Agency | Regression rate study in a small hybrid rocket engine using N2O/paraffin propellant<br>K. BOUGHABA - Université Libre de Bruxelles   | Modeling of the internal hydrogen embrittlement effects on the fatigue life of a structure using a commercial F.E. program<br>W. BOUJILA - DLR |



SP2012 PRELIMINARY PROGRAMME

Wednesday 9th May 2012

| Room          | SC1   | SC2  | SC3  | ST1  | ST2   | Amphi C   |
|---------------|---|--|--|--|---|---|
|               | Session 37- SC - Electric Propulsion (Field Emission Thrusters)   | Session 38 - SC - Propulsion for Cubesat and Nanosat Applications and Studies  | Session 39 - SC - Financial and Operational Related Aspects  | Session 40 - ST - Liquid propulsion Combustion 2   | Session 41 - ST - Hybrid Propulsion 3   | Session 42 - ST - Liquid Engines 1  |
| Chairmen      | S.Pottinger (University of Southampton)<br>D.Estublier (ESA)  | F.Wilson (Aerojet)<br>D.Di Cara (ESA)  | Z.Zuckerman (Rafael)<br>F.Valencia Bel (ESA)   | G. Ordonneau (Onera)<br>M. Valorani (University of Rome)   | O. Orlandi (SAFRAN SME)<br>T. Shimada (Jaxa)  | P. Alliot (SPS)<br>A. Lekeux (Cnes)   |
| 17:20 - 17:40 | FEEP Thruster characterization as a function of the emitter hydraulic resistance<br>L. PAITA - Alta SPA   | Propulsion module for LEO smallsats<br>P. SMITH - Ampac-ISP  | Evolving modularity approach to space propulsion solutions, creates opportunities to shorten design time and reducing cost & risk<br>S. ADLER - Rafael | CFD Simulation of the GCH4/GO2 Mascotte test bench<br>P. GRENARD - ONERA   | Lab-scale motor firing results on advanced solid fuels for hybrid propulsion<br>C. CARMICINO - University of Naples Federico II | Transient simulation of the RL-10A-3-3A rocket engine<br>F. DI MATTEO - University of Rome  |
| 17:40 - 18:00 | Surface electrical resistivity of randomly caesium contaminated FEEP ceramic<br>L. PAITA - Alta SPA   | Desorbitation of the nanosatellite robusta<br>B. PEEV - University Pierre and Marie Curie  | Making the case for <del>WPT</del> versus toxic propellant selections<br>C. JOHNSON - NASA   | Numerical simulation of trans-critical diffusion flame in a rocket engine model combustor<br>Y. MIZOBUCHI - Japan Aerospace Exploration Agency   | Test results from small-scale hybrid rocket testing<br>JER RONNINGEN - Nammo Raufoss  | Evaluation of the potential application of a multiple use turbo machine starter in a gas generator cycle rocket engine<br>W.H. KITSCHKE - German Aerospace Center |
| 18:00 - 18:20 | Time of flight measurement on the plume of the Cs slit FEEP thruster<br>L. PAITA - Alta SPA   | Advanced fluidic equipment and components for micro and small propulsion applications: review of the development, qualification and flight hw manufacturing/acceptance activities at TAS-1<br>G. MATTICARI - Thales Alenia Space | A strategy to minimize dynamic residuals during in-orbit operations of bipicolombo dual-mode propulsion system<br>L. BISI - Astrium                    | Redundant ignition system development overview for the VINCI engine<br>W. KWAN - Aerospace Propulsion Products                                   | Solid modulated propulsion for space exploration<br>P. YVART - SME Safran   | Selection of the propellant of a multiple use turbo machine starter in a cryogenic gas generator cycle rocket engine<br>W.H. KITSCHKE - German Aerospace Center   |
| 18:20 - 18:40 | Ionic liquid FEEP beam characterization<br>S. MARCUCCIO - Alta SPA  | Micropropulsion system with closed-loop regulated thrust for CubeSats<br>H. JOHANSSON - NanoSpace  | Results of the cool gas generator test on board of proba-2 and development towards future space applications<br>B. SANDERS - CGG Technologies          | Experimental and numerical analysis of ignition in a GO2/GCH4 Co-axial injector at ambient and low pressures<br>M. VALORANI - University of Rome | CFD Simulation of liquid injection in hybrid rockets<br>M. LAZZARIN - CISAS, University of Padova                               | Influence of liquid oxygen environment on the tribological behaviour of PCTFE/440C steel contact<br>CAUTAIN - SUPMECA - LISMMA                                    |
|               | <b>16:00 - 18:30 Green Propulsion Workshop - European Working Group Session</b><br>Presentation and discussion on REACH, Clean Space Initiative and key topics from the European Space Technology Harmonisation process |  |  |  |   |   |
| 20:30 - 00:00 | <b>Gala Dinner - Palais de la Bourse - 19 Palais de la Bourse, Bordeaux</b>   |  |  |  |   |   |

SP2012 PRELIMINARY PROGRAMME

Thursday 10th May 2012

| Room          | SC1   | SC2  | SC3   | ST1   | ST2   | Amphi C   |
|---------------|---|--|---|---|---|---|
|               | Session 43 - SC - Electric Propulsion Systems (Electrothermal Thrusters and Studies)  | Session 44 - SC - Propulsion Components - Chemical Propulsion  | Session 45 - SC - Nuclear Propulsion  | Session 46 - ST - Liquid engines 2  | Session 47 - ST - Hybrid Propulsion 3   | Session 48 - Exploration  |
| Chairmen      | H. Neumann (IOM Leipzig)<br>M. Auweter-Kurtz (Stuttgart University)   | P. Smith (Ampac)<br>K. Kajiwara (Jaxa)   | G. Schmidt (NASA)<br>G. Saccoccia (ESA)   | J. Breteau (ESA)<br>D. Chaves (Moog)  | J. Ronningen (Nammo)<br>A. Dellillis (ASI)  | Y. Traissac (SAFRAN-SME)<br>H. Ellerbrock (Astrium)   |
| 09:00 - 09:20 | Development of an annular-geometry ion engine<br>M. PATTERSON - NASA  | A porous metal electrospray array for space propulsion applications<br>C. COFFMAN - Massachusetts Institute of Technology        | Overview of ISTC funded projects in space propulsion area<br>T. RYZHOVA - International Science & Technology Center | Thermochemical study of liquid propellant combinations for space applications<br>R. AMRI - Centre of Space Techniques   | The Patterns of streamwise vortex on the fuel surface in hybrid rocket combustion<br>C. LEE - Konkuk University     | Appraisal of initiated ESA propulsion developments for exploration missions<br>H. ELLERBROCK - Astrium                  |
| 09:20 - 09:40 | The design, development, manufacture and test of an electric propulsion system for small spacecraft<br>M. POLLARD - Surrey Satellite Technology | Basic design of demise propellant tank upon re-entry<br>K. KAJIWARA - JAXA   | Modular aneutronic fusion engine<br>Dr. G. PAJER - Princeton Satellite Systems                                      | Preliminary study on a throttleable 1.5-6 kN MMH/NTO European medium thrust engine<br>R. ARNOLD - Astrium   | Injection effects in hybrid rocket engines<br>M. KOBALD - Deutsches Zentrum für Luft- und Raumfahrt e.V.            | Propulsion system for the European lunar lander, development status and breadboarding activities<br>M. RIEHLE - Astrium |
| 09:40 - 10:00 | A new orbit control algorithm for power processing control unit of a 20mN class ion thruster<br>H. OSUGA - Mitsubishi Electric Corporation      | Development and qualification status of Astrium's new non-ITAR bipropellant 10N thruster flow control valve<br>M. WOLF - Astrium | Nuclear thermal propulsion for advanced space exploration<br>M. HOUTS - NASA  | Technical data and development status of the RD861K engine<br>V. SHULGA - Yuzhnoye SDO  | Through a 1D modeling of liquefying fuel regression rates in hybrid rockets<br>J.Y. LESTRADE - CNES                 | Overview of the in-space propulsion (ISP-1) project<br>P. ALLIOT - Snecma   |
| 10:00 - 10:20 | Combining low-thrust and manifold dynamics for vertical lyapunov orbits missions<br>C. FINOCCHIETTI - University of Pisa                        | Vacuum-arc plant and technology for deposition of MoSi <sub>2</sub> -NbSi <sub>2</sub> coatings<br>V. NADTOKA - Yuzhnoye SDO     | High-power thermal simulation development for non-nuclear testing of nuclear systems<br>J. PEARSON - NASA           | Two-component liquid-propellant low-thrust propulsion system for third stage of Cyclone-4 LV<br>Y. SHOVKOPLYAS - Yuzhnoye SDO                                     | Multi-physics numerical simulation of combustion processes in hybrid rockets<br>A. MAZZETTI - Politecnico di Milano | End-to-end flight test for planetary landing, an unprecedented approach in Europe<br>G. PARISENTI - Thales Alenia Space |
| 10:20 - 10:40 |   | Modelling of fluid motion in spacecraft propellant tanks - Sloshing<br>Dr N. FRIES - Astrium                                     |   | Results of first phase of qualification test of two-component liquid-propellant low-thrust engine for third stage of Cyclone-4 LV<br>V. DURACHENKO - Yuzhnoye SDO | Fuel combustion modelling in hybrid engine<br>M. PREVOST - ONERA  |   |
| 10:40 - 11:10 | Coffee Break  |  |   |   |   |   |

## SP2012 PRELIMINARY PROGRAMME

### Thursday 10th May 2012

| Room          | SC1   | SC2  | SC3  | ST1  | ST2   | Amphi C   |
|---------------|---|--|--|--|---|---|
|               | <b>Session 49 - SC - Electric Propulsion (HEMPT and Field Emission Thrusters)</b>   | <b>Session 50 - SC - Propulsion Components - General</b>   | <b>Session 51 - SC - AIV Issues - General</b>  | <b>Session 52 - ST - Liquid engines 3</b>  | <b>Session 53 - ST - Solid propulsion 4</b>   | <b>Session 54 - ST - Launcher's Attitude Control Systems 2</b>  |
| Chairmen      | N.Puetmann (DLR)<br>V.Ruby (Busek)  | M. Lyszyk (Thales Alenia)<br>C.Hunter (ESA)  | R.Blott (Partnership Ltd.)<br>A.Built (ESA)  | K. Schäffer (DLR)<br>F. Jean (Snecma)  | J. Thepenier (SAFRAN-SME)<br>J. Gigou (ESA)   | N. Girard (CNES)<br>P. Yvart (SAFRAN-SME)   |
| 11:10 - 11:30 | Alta's FT-150 FEFP - overview and development status<br>L. PAITA - Alta SPA   | Electronic pressure regulator development and test results<br>P. SMITH - Ampac-ISP   | Determination of Xe Ion velocities by Laser induced fluorescence in the near field plasma of a small power Ukrainian HET Thruster<br>T. GIBERT - GREMI | System engineering of the European stage combustion demonstrator score-d<br>P. DANOUS - SNECMA   | BOOST TP - A new step in high performance thermoplastic composite structure<br>B. DEFOORT - Astrium   | Development of a cold gas attitude control system for a lander demonstrator<br>K. ODIC - Astrium  |
| 11:30 - 11:50 | Alta's FEFP glass emitter design, manufacturing and test characterization<br>L. PAITA - Alta SPA  | Parametric study of spacecraft plug cluster nozzle<br>I. MASUDA - JAXA   | Propulsion Lab (EPL)<br>J. GONZALEZ DEL AMO - ESA  | From paper to production to test : an update on NASA's J-2X engine for exploration<br>H. KYNARD - NASA   | Two-phase flow modeling to simulate alumina droplet impingements on solid rocket motor (SRM) nozzle<br>V. FROMENT - SPS                                   | Testing of a liquid oxygen/liquid methane reaction control thruster in a new altitude rocket engine test facility<br>M. MEYER - NASA  |
| 11:50 - 12:10 | Overview, qualification and delivery status of the HEMPT based ion propulsion system for SmallGEO<br>S. WEIS - Thales Electron Devices                                    | Generic fill and drain valve delta-qualification<br>D. AVERY - Ampac-ISP   | New developments related to the advanced electric propulsion diagnostic system<br>H. NEUMANN - IOM   | An End-to-End High Fidelity Numerical Simulation of the LE-X Engine - Innovative Evaluation Methodology of Combustion Chamber Hazard -<br>N. TANI - JAXA | Ensuring solid propellants long term availability for spacecrafts and launchers<br>L. CHAMBRAS LAFUENTE - SME   | Advanced solid propulsion technologies for attitude control systems<br>P. CAUBET - SPS  |
| 12:10 - 12:30 | Development and qualification status of the HEMP thruster and the hollow cathode neutralizer for the ion propulsion system on SmallGEO<br>M. SCHIRRA - Thales Deutschland | European latch valve developments<br>R. MATTHIJSSEN - Bradford Engineering   | In-situ EP thruster characterization with a thermocamera<br>H. NEUMANN - IOM   | Experimental investigation of brazilian liquid rocket engine<br>W.F. ALVES - Institute of Aeronautics and Space  | Twin screw process demonstration technology activities for new generation launcher applications<br>S. SAINT MARTIN - SME-Safran                           | Studies of the processes of gasification liquid propellant for designing the rocket engine on scheme "gas-gas" for autonomous on-board deorbiting system of separating part stages of space launcher vehicle on orbits of the utilization<br>V. TRUSHLYAKOV - OmSTU |
| 12:30 - 12:50 | Progress in the endurance testing of HEMPT ion propulsion module for SmallGEO<br>A. LAZURENKO - Thales Electron Devices   | Status of pyrovalve manufacturing at EADS Astrium and assessment of design evolutions for future spacecraft needs<br>M. WOLF - Astrium | XPS Plasma propulsion system on AlphaBus & AlphaSat<br>M. LYSZYK - Thales Alenia Space   | Manufacturing and testing status of the LM10-MIRA LOX-LNG demonstrator<br>L. ARIONE - AVIO   | Verification of image processing method for X-ray CT image to obtain orientation data of AP particles<br>K. KITAGAWA - Japan Aerospace Exploration Agency | On the development and successful application of a thrust modulation module for the FUSEX hybrid rocket motor<br>J. HIJLKEMA - ONERA  |
| 12:50 - 14:30 | Lunch   |  |  |  |   |   |

SP2012 PRELIMINARY PROGRAMME

Thursday 10th May 2012

| Room          | SC1  | SC2  | SC3   | ST1   | ST2  | Amphi C   |
|---------------|--|--|---|---|--|---|
|               | Session 55 - SC - Electric Propulsion (Ion and Hall Effect Thrusters)  | Session 56 - SC - 3. Production and Manufacturing Issues   | Session 57 - SC - New and Green Propellants in Chemical Propulsion  | Session 58 - ST - Liquid Propulsion Stages 2  | Session 59 - Future Propulsion   | Session 60 - 11. Technology Building Blocks for Future Spacecraft Propulsion Systems                  |
| Chairmen      | C.Koppel (Kopoos Consulting)<br>G.Popov (Riame)  | R.Killinger (Astrium)<br>M.Wolf (Astrium)  | I.Coxhill (Ampac-ISP)<br>C. Kappenstein (LACCO)   | M. Meyer (NASA)<br>P. Fortunier (CNES)  | JF Guery (SAFRAN-SME)<br>C. Lardier (Air & Cosmos)   | G.Schmidt (NASA)<br>Manesse (Snecma) N.   |
| 14:30 - 14:50 | Electric propulsion system design impacts resulting from dual thruster operations for the BepiColombo mission<br>H. GRAY - Astrium | Additive layer manufacturing for integrated propulsion systems<br>S. BARLEY - EADS Innovation Works                | Low cost production and characterization of hydroxylammonium nitrate (HAN) monopropellant<br>K.S. KOH - The University of Nottingham Malaysia Campus  | Choice of Helium storage for A5ME upper stage<br>F. MASSON - CNES   | E-Plus: Electric Propulsion equipped Launchers Upper Stage<br>P. PERGOLA - Alta SPA  | High power nuclear electric propulsion roadmap<br>R.J. BLOTT - Space Enterprise Partnerships          |
| 14:50 - 15:10 | Autonomous operation of the electric propulsion system for the BepiColombo mission<br>H. GRAY - Astrium                            | Industrial organisation of propulsion subsystem for satellites constellation<br>M. BATTISTEL - Thales Alenia Space | Overview of the development of ADN-based propellants<br>N. WINGBORG - FOI, Swedish Defence Research Agency  | Heat exchange and pressure drop in a sloshed tank<br>T. HIMENO - University of Tokyo                                      | Development of continuous detonation wave combustion for space launcher<br>J. FALEMPIN - MBDA  | Electric propulsion transit analyses for short stay Mars missions<br>J. DANKANICH - NASA              |
| 15:10 - 15:30 | Performance characterisation of the T6 gridded ion thruster for BepiColombo<br>A. GRUBISIC - Flux Engineering                      | Additive manufacturing advantages for space propulsion<br>M. SMITH - ESA-ESTEC                                     | Thruster on the oxyhydrogen composition obtained directly on the board of satellite from water<br>V.A. MENSHIKOV - International Committee on IGMASS Project Implementation (ICPI)              | Cryogenic propellant storage and transfer technology demonstration for long duration in-space missions<br>M. MEYER - NASA | Notional roadmap for developing nuclear propulsion systems for human space exploration<br>U. KAMATH - Boeing                                 | Benefits of the annular gridded-ion engine<br>J. DANKANICH - NASA                                     |
| 15:30 - 15:50 | Hollow cathode thruster characterisation at low flow rates<br>D. LAMPROU - University of Surrey                                    | Combustion chamber design for additive manufacturing<br>S. HYDE - ESA  | The structural effect of substituted barium hexaaluminate catalysts on the decomposition of N2O as a propellant<br>T. ZHANG - Dalian Institute of Chemical Physics, Chinese Academy of Sciences | Depressurisation tests of cryogenic liquid in low gravity environment<br>J. LACAPERRE - Air Liquide                       | Trajectory and cost analysis of microwave rocket launches<br>A. ARNAULT - University of Tokyo, Ecole Centrale Paris                          | High power electric propulsion for a flexible space exploration architecture<br>P. PERGOLA - Alta SPA |
| 15:50 - 16:10 | HT100: Cathode coupling test and endurance<br>D. DIGNANI - Alta SPA  | A design optimisation study of a generic bi-propellant injector for additive manufacturing<br>S. HYDE - ESA        | Research and development of HAN (HydroxylAmmonium Nitrate) based monopropellant thruster<br>N. AZUMA - JAXA   | New key technologies matured for the A5ME and next generation launchers<br>S. BIANCHI - Air Liquide                       | Zooming capabilities of the 1D ESPSS propulsion simulation tool with 3D-CFD solvers : implementation and validation<br>J. PRAUSE - ESA-ESTEC | Propulsion options for fast HERRO missions to Mars<br>R. GEORGE SCHMIDT - NASA                        |
| 16:10 - 16:40 | Coffee Break   |  |   |   |  |   |

## SP2012 PRELIMINARY PROGRAMME

### Thursday 10th May 2012

| Room          | SC1   | SC2  | SC3  | ST1  | ST2   | Amphi C  |
|---------------|---|--|--|--|---|--|
|               | <b>Session 61 - SC - Electric Propulsion (MPD Thrusters and Studies)</b>  | <b>Session 62 - SC - Electric Propulsion (Studies)</b>   | <b>Session 63 - SC - Green Propulsion for Spacecraft - Hydrogen Peroxide</b>   | <b>Session 64 - ST- Liquid Propulsion High Frequency Unstabilities</b>   | <b>Session 65 - ST - Liquid Propulsion Testing</b>  | <b>Session 66 - Theoretical Performance vs Status of Technology: How to Push the Limits + Future Propulsion</b>  |
| Chairmen      | M.Coletti (University of Southampton)<br>J.Marcos (Technalia)   | J.Polk (JPL)<br>H.Leiter (Astrium)   | A.Musker (Deltacat)<br>L.D'Agostino (Alta)   | O. Haidn (DLR)<br>P. Danous (Snecma)   | F. Masson (CNES)<br>S. Bianchi (Air Liquide)  | S. Henry (SAFRAN-SME)<br>A.Demaire (OHB)   |
| 17:00 - 17:20 | Performance characterization of a 100-kW-class applied-field MPD thruster<br>R. ALBERTONI - Alta SPA  | Superconductor coils to generate the magnetic field of a hall effect thruster<br>J. IBARZO - Technalia                                 | A study of ignition delay for a 300 N hydrogen peroxide bipropellant thruster<br>A. MUSKER - Deltacat                                    | Acoustic Energy Content Investigated as a Function of Operational Conditions for LOX/GH2 with and without External Forcing<br><del>W. WEBSTER - DLR</del>      | Ground testing for upper stage qualification<br>K. SCHÄFER - DLR  | Identification of thrust components in the Helicon Double Layer Thruster<br>C. CHARLES - Australian National University  |
| 17:20 - 17:40 | Insert temperature measurements of a 180A hollow cathode for the HIPER Project<br>M. COLETTI - University of Southampton                                    | 3-Dimensional mapping of the plasma properties in the plume region of the quad confinement thruster<br>A. KNOLL - University of Surrey | Development model of a green 1N Bi-propellant thruster for attitude control<br>A. WOSCHNAK - Fotec                                       | Experimental study of the combustion-acoustics coupling in liquid rocket engine high-frequency instabilities<br>Y. MERY - Snecma                               | Pump fed hydrogen peroxide rocket propulsion testing<br>S. MEYER - Purdue University  | Attitude and orbit control systems for GEO bird only based on electric propulsion<br>A. DEMAIRE - OHB Sweden   |
| 17:40 - 18:00 | Thrust stand performance measurements of a high-power gallium electromagnetic thruster<br>R. THOMAS - NASA  | Plasma propulsion with electronegative gases. A status update<br>A. AANESLAND - CNRS-Ecole Polytechnique                               | Development of a hydrogen peroxide/ethanol thruster for the advanced re-entry vehicle<br>W.P.W. WIELING - Delft University of Technology | Response of a combusting LOx jet to a transverse acoustic oscillation<br>J. HARDI - DLR  | Experimental investigation of phase-change behaviour in a water-ice heat accumulator setup for cryogenic application<br>J. RICCIUS - DLR                              | Design of a new liquid propellant for space launchers : synthesis, process and propulsive performances<br>A.J. BOUGRINE - Laboratoire Hydrazines et Composés Energétiques Polyazotés |
| 18:00 - 18:20 | Cold start cathode with wide range of operation<br>A. LOYAN - National Aerospace University   | Solar electric propulsion vehicle demonstration to support future space exploration missions<br>K. BRYAN SMITH - NASA                  | Performance of a 2N microthruster using hydrogen peroxide and a mixed oxide bulk catalyst<br>R. VIEIRA - INPE                            | Numerical and analytical study on the acoustic structure in a cylindrical rocket chamber during oscillation<br>T. SHIMIZU - Japan Aerospace Exploration Agency | Steady-state and transient simulation of test bench flow phenomena<br>R. BEHR - Astrium   | FAST20XX Hybrid Propulsion Technologies: Objectives and Status<br>R. MOLINA - ESA/ESTEC  |
| 18:20 - 18:40 | A new flexible magnetic circuit for a hall thruster : results from the first campaign in the Pivoine-2G ground-test facility<br>L. GARRIGUES - Laplace-CNRS | Comparison of computed and measured performance of a pulsed inductive thruster operating on argon propellant<br>K. POLZIN - NASA       | Peroxy-fuels for rocket propulsion<br>K.B. MISHRA - BAM Federal Institute for Materials Research and Testing                             | Development of an advanced global MMH-NTO combustion model for CFD application<br>T. AICHNER - Astrium   | Combustion stability verification of the Vega LV upper stage main engine at free gas injection into the engine channels<br>V. SHNYAKIN - Yuzhnoye State Design Office |  |

END OF CONFERENCE